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Influence of Vortex Induced Loads on the Motion of SPAR-Type Wind Turbine: A Coupled Aero-H
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1 **Influence of Vortex Induced Loads on the Motion of SPAR-Type Wind**
2 **Turbine: A Coupled Aero-Hydro-Vortex-Mooring Investigation**

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38 **Abstract**

39

40 *The nonlinear coupling effect between DOFs and the influence of vortex induced loads*
41 *on the motion of SPAR type FOWT are studied based on an aero-hydro-vortex-mooring*
42 *coupled model. Both first- and second-order wave loads are calculated based on the 3D*
43 *potential theory. The aerodynamic loads on the rotor are acquired with the blade element*
44 *momentum theory. The vortex induced loads are simulated with CFD approach. The*
45 *mooring forces are solved by the catenary theory and the nonlinear stiffness provided by*
46 *the SPAR buoy are also considered. The coupled model is set up and a numerical code is*
47 *developed for calculating the dynamic response of a Hywind SPAR-type FOWT under the*
48 *combined sea states of wind, wave and current. It shows that the amplitudes of sway and*
49 *roll are dominated by lift loads induced by vortex shedding, and the oscillations in roll*
50 *reach the same level of pitch in some scenarios. The mean value of surge is changed*
51 *under the drag loads, but the mean position in pitch, as well as the oscillations in surge*
52 *and pitch, is little affected by the current. Due to the coupling effects, the heave motion is*
53 *also influenced by vortex-induced forces. When vortex-shedding frequency is close to the*
54 *natural frequency in roll, the motions are increased. Due to nonlinear stiffness,*
55 *super-harmonic response occurs in heave, which may lead to internal resonance.*

56

57 **Keywords:** *floating offshore wind turbines, coupled model, current, vortex induced*
58 *motion, internal resonance, nonlinear stiffness, super-harmonic*

59

60 **1. Introduction**

61 With growing demands, the problem of energy shortage attracts more and more
62 attention all over the world. Among different technologies, wind turbines convert wind
63 energy into electricity with no pollution or waste [1]. Because of the greater intensity and
64 stability of the offshore wind, the capacity of offshore wind turbine usually surpasses that
65 of the onshore ones [2,3]. With the development of offshore technology, the research on
66 offshore wind turbines is moving towards the deep-water zone. Compared with the fixed
67 offshore wind turbine, the advantage of the floating offshore wind turbine (FOWT) is
68 pronounced in the aspects of economy, convenience of installation and total capacity [4].

69 At present, the design of FOWT is based on experiences from offshore oil and gas
70 production platforms [5]. Based on the floating foundations, FOWTs can be divided into
71 barge, SPAR, semi-submersible and tension-leg-platform (TLP) types. During the past
72 decades, full-scale prototypes of FOWT have been successfully launched and tested all
73 over the world, greatly expediting the development of FOWT technologies [6]. Among
74 those designs, the SPAR-based wind turbine tethered by multiple cables shows robust

75 hydrodynamic performance [7].

76 With incoming currents, vortex induced vibration (VIV) is an important source of
77 disturbance on offshore structures such as risers, pipelines [8-12], and FOWTs. When the
78 vortex induced loads act on a rigid body with long and round shape but also large
79 displacement, it will hardly cause any structural vibration but may induce additional
80 motion in some degree of freedom (DOF) under specific conditions. This phenomenon is
81 called Vortex Induced Motions (VIM). Maija and Benitz[13] studied the dynamic
82 response of DeepCWind semi-submersible FOWT with incoming currents based on
83 OpenFOAM. They found that the vortex shedding would cause large time-varying load,
84 which affects the fatigue life of the system. Kokubun et.al. [14] conducted a 1/34.5 scaled
85 model test, and recorded VIM frequency in sway, roll and mooring tension. Duan et. al.
86 [15] performed model testing with various current, wind and wave conditions. The
87 lock-in phenomenon of sway in the cross-flow direction was observed and the remaining
88 responses, including the other 4-DOF motions, mooring tensions, and turbine bearing
89 loads, were found to be coupled *via* sway/surge VIMs.

90 In previous studies, CFD approaches have been widely adopted in VIV investigations
91 with slender risers. Li et.al [39] employed a partitioned iterative scheme based on Petrov–
92 Galerkin formulation to simulate the VIV of an elastically mounted circular cylinder with
93 2D and 3D models, in which the wall proximity effects were observed. Mitta [40]
94 examined the VIV of a circular cylinder with a stabilized space–time finite element
95 formulation and identified three branches in the response. Bourguet et. al [41]
96 investigated the multi-frequency VIV of a cylindrical tensioned beam under the scenario
97 of shear flows. They found that the structural responses were determined by the shape of
98 inflow profile. Wang et.al proposed a 3D fluid-structure interaction model to simulate the
99 2DOF VIV characters of a vertical riser [42,43], and observed different vortex shedding
100 modes. Furthermore, a model was developed to simulate the couple VIV effect of two
101 tandem flexible cylinders [44].

102 The hydrodynamic significance of VIM was widely studied in slender structures
103 including the SPAR platform with both numerical methods and experimental approaches.
104 Hirabayashi[16] numerically analyzed the VIM of 2D circular cylinders by using the
105 lattice Boltzmann method, and the changing trends of lift load were observed in his work.
106 Wu et.al [17] employed OpenFOAM to investigate the free vibration of a square cylinder
107 in transient flow with three hybrid turbulence models. A 3D model was established in
108 their simulation. The good agreement between their results and experimental data proves
109 that the CFD approach is accurate enough to handle the vortex shedding problem for the
110 VIM of a slender body.

111 Meanwhile, more work has been performed on the VIM of semi-submersible buoys,
112 another group of slender structures. Hashiura et.al [18] conducted a series of towing
113 experiments in water tank to investigate the relationship between the vortex induced force
114 and the shape of buoy, as well as other parameters. Similarly, Liu et.al [19] carried out a
115 group of model tests aiming at understanding the fluid physics associated with VIMs of
116 deep-draft semi-submersibles. They found the wake behind the pontoons has
117 non-negligible influence on the dynamic behavior of the buoy. Based on their
118 experimental work, Liang et.al [20] further established 3D numerical models to simulate
119 the vortex shedding in the wake as well as its effect on the motion of the rigid body.

120 Among the researches on SPAR platform, the wave-frequency motion and VIM are
121 usually studied independently. In order to consider these effects simultaneously, Liu [21]
122 created a coupled model for SPAR platform under the combined action of wave and
123 vortex shedding caused by current. In his study, a 3-DOF model is developed to simulate
124 the heave, roll and pitch of a SPAR. First-order wave force, second-order wave force and
125 vortex induced force are considered in his numerical model, but the influence of mooring
126 line is not included. Meanwhile, there are few researches on the SPAR-type floating wind
127 turbine under the combined environmental loads of wave, wind and current.

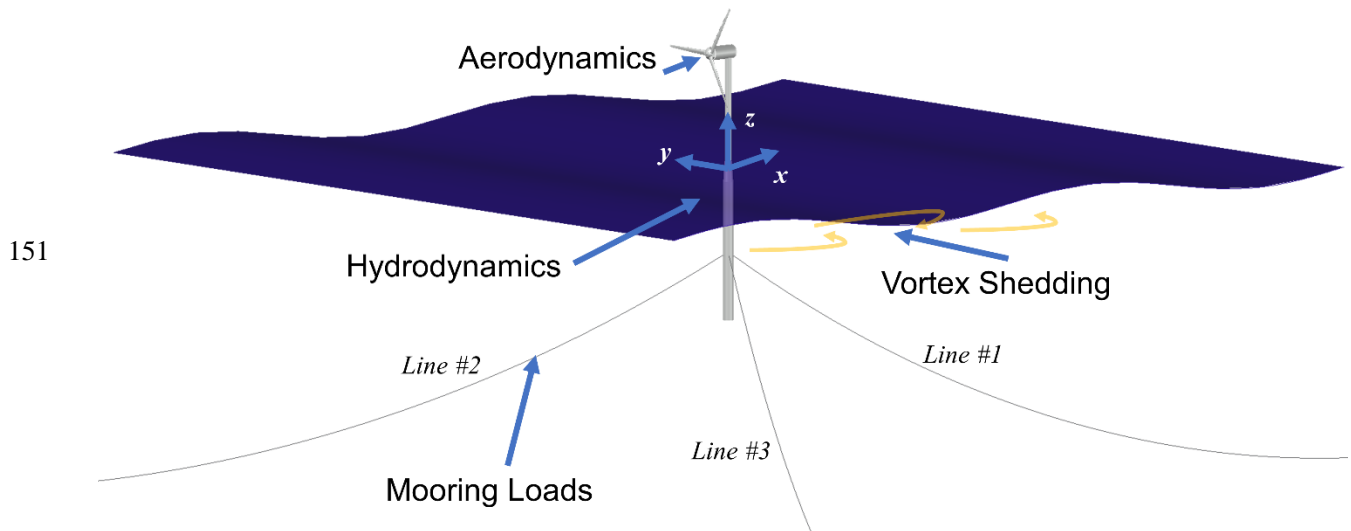
128 In the present work, an in-house coupled model for SPAR-type FOWT is developed.
129 Based on the potential flow theory, both first-order and second-order difference
130 frequency wave forces are calculated with stochastic waves. The aerodynamic load on the
131 wind turbine is calculated by the blade element momentum theory (BEM). The
132 hydrodynamic coefficients of vortex are calculated by CFD approach. By coupling these
133 modules in time domain, this method is capable of analyzing the dynamic response of
134 SPAR-type FOWT under complex sea states.

135 In the following sections, the physical problem, including the configuration as well as
136 physical parameters of the floating wind turbine system, is defined firstly. Afterwards, the
137 numerical models (including the nonlinear restoring forces model, the catenary mooring
138 model, the aerodynamics model, the vortex induced force model and the hydrodynamics
139 model) are briefly described. Numerical results, including predictions of dynamic
140 responses under wave, wind and current, are then presented. Finally, conclusions are
141 drawn.

142 **2. Physical problem**

143 As shown in Fig. 1, the FOWT studied in this work consists of the NREL 5MW
144 baseline wind turbine [22] (see Table 1) and a SPAR-type floating foundation with three
145 mooring lines [23] (see Table 2). The mooring cables are located around the buoy body of
146 the SPAR. One of the cables (Line #1) is directed along the positive x -axis in the xz -plane,

147 and the other twolines (Line #2 and #3) are distributed uniformly around the platform.
 148 Hereby (x,y,z) is a Cartesian coordinate system with its origin at the mean free surface
 149 and z pointing upward. The x axis coincides with the direction of the incoming wind,
 150 wave and current.



152 Fig.1 Definition of the physical problem

153

154 Table 1 Parameters of the NREL 5MW wind turbine

Parameter	Value
Rated power	5 MW
Shaft transmission efficiency	0.944
Radius of wind wheel	63 m
Radius of hub	1.5 m
Cut-in wind speed	3 m/s
Rated wind speed	11.4 m/s
Cut-out wind speed	25 m/s
Rated speed	12.1 rpm
Hub height(from the bottom of the tower)	90 m
CM location(from the bottom of the tower)	64.0 m
Total mass(including tower)	697,460 kg

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Table 2 Parameters of the SPAR-type floating foundation and the mooring system

Parameter	Value
Depth to platform base below the SWL	120.0 m
Elevation to platform top above the SWL	10.0 m
Depth to top of taper below the SWL	4.0 m
Depth to bottom of taper below the SWL	12 m
Platform diameter above taper	6.5 m
Platform diameter below taper	9.4 m
Platform mass, including ballast	7,466,330 kg
CM location below the SWL along platform centerline	89.9155 m
Number of mooring lines	3
Angle between adjacent lines	120 deg
Depth to anchors below SWL (water depth)	320 m
Depth to fairleads below the SWL	70 m
Radius to anchors from the platform centerline	853.87 m
Radius to fairleads from the platform centerline	5.2 m
Unstretched mooring line length	902.2 m
Mooring line diameter	0.09 m

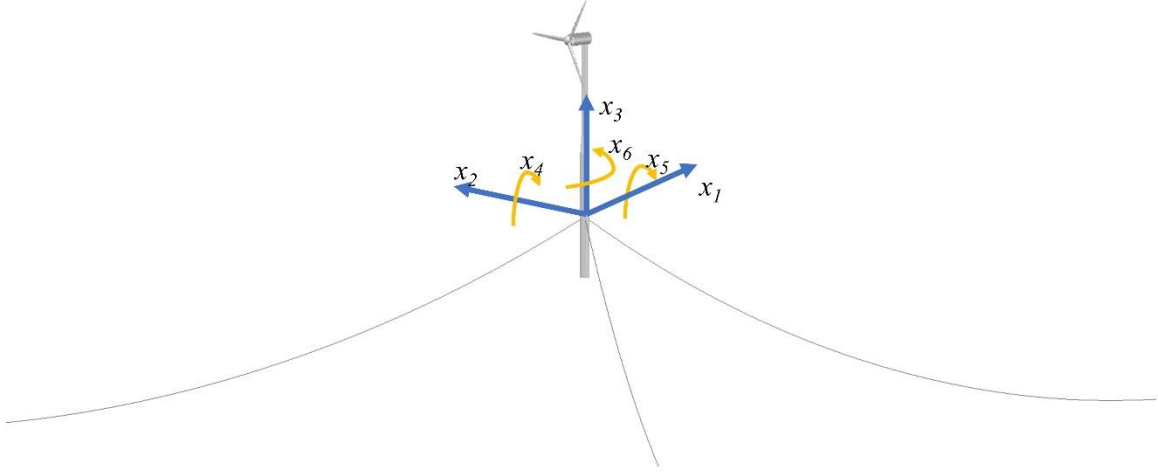
157 3. Methodology

158 3.1 Dynamic equation in time domain

159 The dynamic equation is developed to calculate the displacement, velocity and
160 acceleration of FOWT in time domain. For moored floating offshore structures, the
161 dynamic response is solved under the effects of wind, wave, current and cable forces.
162 Considering the 6-DOF motions (see Fig.2), the governing equation can be written in the
163 time domain as following,

$$164 \quad [M + A(\omega)]\ddot{x} + C(\omega)\dot{x} + Df(\dot{x}) + K(x)x = q(t, x, \dot{x}), \quad (1)$$

165 where M is the body mass and inertia matrix, A is the frequency-dependent added mass
166 matrix, and C is the frequency-dependent radiation damping matrix. D is the nonlinear
167 damping matrix. f is the vector function of \dot{x} . K is the restoring matrix provided by
168 buoyancy. x, \dot{x}, \ddot{x} represent the 6-DOF position, velocity and acceleration vectors of the
169 body, respectively. q is the exciting force vector, which includes the first- and
170 second-order wave loads, the nonlinear restoring forces provided by the mooring lines,
171 the vortex induced loads, and the aerodynamic loads on the rotor.



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Fig.2 Six DOFs of FOWT

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The added mass and radiation damping coefficients calculated based on the 3D potential theory are frequency dependent. With irregular waves, it is difficult to choose the corresponding added mass and damping coefficients for the time domain equations. In order to solve this problem, the frequency-dependent added mass and radiation damping coefficient are transferred into the added mass corresponding to the infinite frequency and retardation function based on the convolutional method [24]. Thus, the governing equation can be written as,

181

$$(\mathbf{M} + \mathbf{A}_\infty) \ddot{\mathbf{x}}(t) + \int_0^t \mathbf{h}(t-\tau) \dot{\mathbf{x}}(\tau) d\tau + \mathbf{Df}(\dot{\mathbf{x}}) + \mathbf{K}(\mathbf{x})\mathbf{x} = \mathbf{q}(t, \mathbf{x}, \dot{\mathbf{x}}), \quad (2)$$

182

183

where $\mathbf{h}(t)$ is the retardation function, and \mathbf{A}_∞ is the added mass when the frequency approaches infinite.

184

3.2 Wave loads

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The wave is assumed to propagate along the positive x-axis, and stochastic wave elevation $\eta(t)$ can be decomposed into the sum of N regular wave components as following,

188

$$\eta(t) = \sum_{n=1}^N a_n \cos(k_n x - \omega_n t + \varphi_n). \quad (3)$$

189

190

191

For each component, a_n denotes the wave amplitude, ω_n is the circular frequency, k_n is the wave number, and φ_n is the random phase angle. The wave amplitude a_n can be calculated by the corresponding wave spectrum S_η .

192 To obtain the wave force on the platform in time domain, the load transfer functions
 193 are calculated based on the 3D potential theory in frequency domain by using the
 194 DNVGL software WADAM. Specifically, the hydrodynamic transfer function includes
 195 linear transfer function (LTF) $F_1(\omega)$, as well as the sum-frequency quadric transfer
 196 function (QTF) $F_{2s}(\omega_i, \omega_j)$ and difference-frequency QTF $F_{2d}(\omega_i, \omega_j)$. Based on our
 197 previous analysis on SPAR-type FOWT [25], the second-order sum-frequency wave load
 198 will not significantly affect the dynamic response of the floating buoy due to its low
 199 natural frequencies. It is thus not included in the following simulations.

200 Afterwards, the random wave forces are transferred into time series by multiplying
 201 these hydrodynamic parameters and specified wave spectrum in the complex domain [26].
 202 The real part of the complex expression will be the corresponding terms of wave loads in
 203 time domain so that we have,

$$204 \quad F_{wave_1}(t) = \text{Re} \left[\sum_{i=1}^M \eta_i F_1(\omega_i) \right] = \text{Re} \left[\sum_{i=1}^M a_i \exp[i(\omega_i t + \phi_i)] F_1(\omega_i) \right], \quad (4)$$

$$205 \quad F_{wave_2d}(t) = \text{Re} \left[\sum_{i=1}^M \sum_{j=1}^M \eta_i \eta_j^* F_{2d}(\omega_i, \omega_j) \right] = \text{Re} \left[\sum_{i=1}^M \sum_{j=1}^M a_i a_j \exp[i((\omega_i - \omega_j)t + \phi_i - \phi_j)] F_{2d}(\omega_i, \omega_j) \right], \quad (5)$$

206 where η_i and η_i^* donate the elevation and its conjugation of i -th wave component in
 207 complex domain. a_i , ω_i and ϕ_i donate the amplitude, frequency and phase of i -th wave
 208 component as mentioned above, respectively.

209 3.3 Vortex induced loads

210 A two-dimensional cylinder model was developed to investigate the vortex shedding
 211 and its induced loads on the SPAR buoy. The CFD package Fluent is adopted to calculate
 212 the lift and drag coefficients, as well as the vortex shedding frequency. Based on these
 213 results, the time varying distributed vortex induced lift and drag forces can be obtained as
 214 below,

$$215 \quad F_L(t) = \frac{1}{2} C_L \rho_c U_\infty^2 D \cos(2\pi f_s t + \alpha), \quad (6)$$

$$216 \quad F_D(t) = \frac{1}{2} C_{Dm} \rho_c U_\infty^2 D + \frac{1}{2} C_{Da} \rho_c U_\infty^2 D \cos(4\pi f_s t + \alpha), \quad (7)$$

217 where C_L and C_{Da} are the amplitudes of the lift and drag coefficients, respectively. C_{Dm} is
 218 the mean value of drag coefficients. ρ_c is the density of the current. U_∞ is the inflow
 219 velocity. D is the diameter of the SPAR buoy. f_s is the vortex shedding frequency, and α is
 220 the phase angle. Specifically, the oscillation frequency of the lift force is the same as the
 221 vortex shedding frequency, while that of the drag force is twice the vortex shedding
 222 frequency. [37]

223 The vortex induced loads act on the wet surface of the SPAR. Thus, the forces and
 224 moments can be calculated by integrating the distributed lift and drag forces along the

225 depth z . We have

$$226 \quad F_{v1}(t) = \int_{-h}^0 F_D(t) dz, \quad (8)$$

$$227 \quad F_{v2}(t) = \int_{-h}^0 F_L(t) dz, \quad (9)$$

$$228 \quad T_{v4}(t) = \int_{-h}^0 F_L(t)(z - z_G) dz, \quad (10)$$

$$229 \quad T_{v5}(t) = \int_{-h}^0 F_D(t)(z - z_G) dz, \quad (11)$$

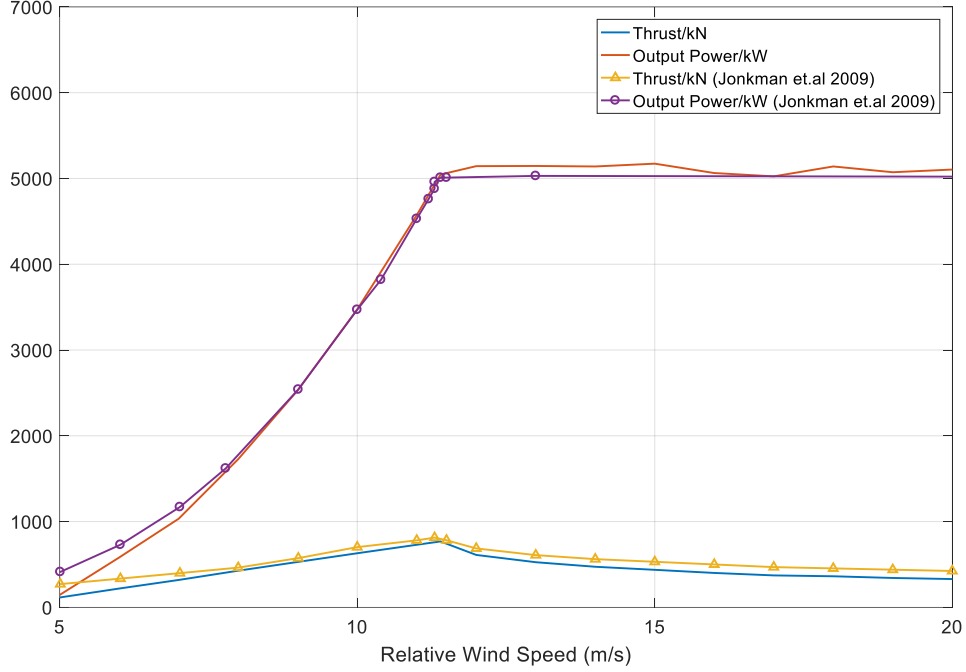
230 where z_g is the center of gravity. $F_{v1}(t)$ and $F_{v2}(t)$ represent the vortex-induced forces
231 in surge and sway respectively, while $T_4(t)$ and $T_5(t)$ represent the vortex induced
232 moments in roll and pitch.

233 3.4 Aerodynamic loads

234 Although it is a quasi-static algorithm, the Bladed Element Momentum (BEM)
235 method has proven to be a simple but accurate way to calculate aerodynamic forces
236 acting on the wind turbine blades, [27,28]. In this work, the BEM approach is adopted to
237 simulate the aerodynamic loads applied on the rotor when the turbine operates.

238 Each blade is discretized into seventeen parts along the span. Within each part the
239 blade elements have the same airfoil shape. The axial induction factor a and tangential
240 induction factor a' can be calculated by iterations at each blade element based on the
241 parameters of the airfoil, such as chord, local pitch angle etc. Then, thrust and torque at
242 each element are determined. After the local aerodynamic loads for all control volumes
243 are obtained, we can get the normal and tangential load distributions. The general thrust
244 and torque on rotor can be acquired by integrating along the span. With these
245 distributions, the aerodynamic performance of the rotor, such as thrust on rotor, power
246 output and bending moment at the root of blade, could be analyzed. A tip loss model, hub
247 loss model and Glauert correction are also adopted to fix the induction factor due to finite
248 blade number vortex shedding from the hub, and turbulent wake. Besides, the
249 motion-induced and vortex-induced velocities of the floating foundation are also
250 considered. Further details on the BEM approach could be found in [29].

251 In this study, the airfoil data of the NREL-5MW wind turbine are adopted [22]. Both
252 the lift and drag coefficients are corrected for rotational stall delay and the drag
253 coefficients are also corrected using the Viterna method. The detailed correction progress
254 could be found in [22]. To validate our aerodynamic model, the thrust on the rotor as well
255 as the output power is calculated. These results are then compared with the data in [22]
256 (Fig.3). It is shown that the accuracy of our model meets the requirement of the
257 simulation.



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Fig.3 Validation of the aerodynamic model

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3.5 Restoring forces

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The nonlinear restoring forces of SPAR-type FOWT is composed of two parts. One part is provided by the mooring system, and the other is the hydrostatic force of the SPAR buoy. A separate module was programmed to calculate the tensions in each cable according to the catenary theory, a quasi-static algorithm to predict mechanics of the mooring system [30]. On the other hand, previous studies on SPAR show that the nonlinear restoring forces of this type floating foundation are mainly reflected in heave, roll and pitch [31,32]. Based on the geometric characteristics of SPAR platform, the additional restoring forces/moments in heave, roll and pitch can be expressed as

269

$$F_3 = \rho g A_w \left(-\eta - \frac{1}{2} H_g x_4^2 - \frac{1}{2} H_g x_5^2 \right), \quad (12)$$

270

$$M_4 = -\frac{1}{2} \rho g \left(\nabla + 2A_w \times \overline{GM}_4 \right) x_3 x_4 + \frac{1}{2} \rho g \left(\nabla + 2A_w \times \overline{GM}_4 \right) \eta x_4, \quad (13)$$

271

$$M_5 = -\frac{1}{2} \rho g \left(\nabla + 2A_w \times \overline{GM}_5 \right) x_3 x_5 + \frac{1}{2} \rho g \left(\nabla + 2A_w \times \overline{GM}_5 \right) \eta x_5, \quad (14)$$

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274

275

where ρ denotes the density of water, g denotes the gravitational acceleration, A_w is the area of water line, ∇ is the displacement volume, \overline{GM}_4 and \overline{GM}_5 are the initial metacentric heights in roll and pitch, respectively, η denotes the elevation of wave, H_g denotes the height of center of gravity, x_3 , x_4 , and x_5 denote the heave, roll and pitch of the

276 SPAR buoy as shown in Fig.2. More details of derivation could be found in Refs. [31,32].
277 In our simulation, the nonlinear restoring load is a part of the external loads \mathbf{q} in Eq. (2).

278 **3.6 Flow chart of simulation**

279 Based on the algorithms above, a coupled aero-hydro-vortex dynamic simulation tool
280 for SPAR-type FOWT is developed in the time domain. The basic procedure of this
281 coupled numerical simulation is shown in Fig. 4.

282 Before the time domain simulation begins, the initial conditions, hydrostatic and
283 hydrodynamic coefficients are pre-generated. At each time step, the motions of COG are
284 numerically calculated from Eq. (2) using the 4th order Runge-Kutta method. For
285 simplicity, structural flexibility is not included in this model so that the wind turbine and
286 floating platform are modeled as a rigid body. The motion of fairlead and rotor could also
287 be calculated. Thus, the right-hand side of Eq. (2) could be updated for each module and
288 integrated to the next time step.

289 Compared with the other existed codes, our simulation tool is specifically developed
290 for the purpose of investigating the motion of SPAR-buoy FOWT under the complex sea
291 states, specifically the VIMs caused by currents. Moreover, we take the nonlinear
292 coupling effect between DOFs of SPAR buoy into consideration, which is usually not
293 considered in other studies.

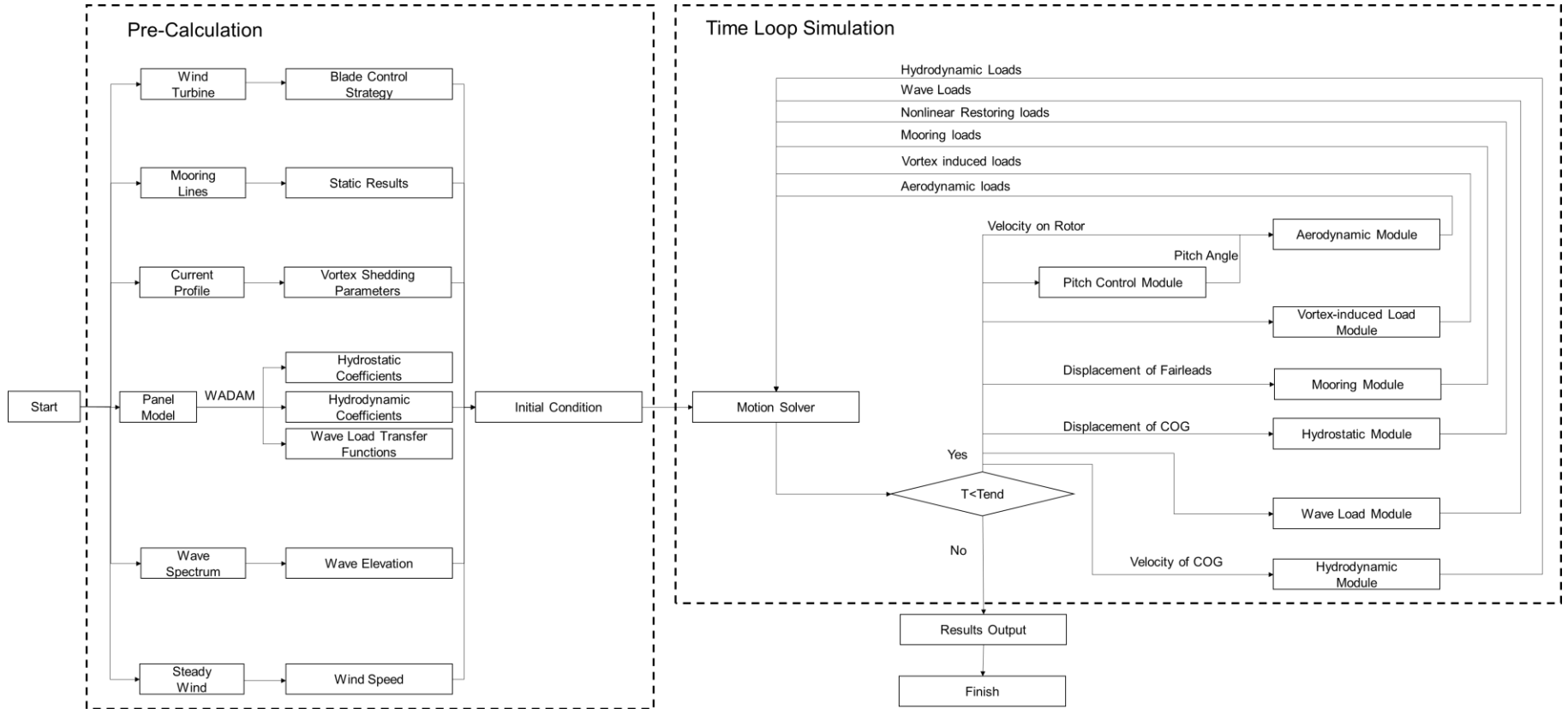


Fig.4 Basic procedure

297 **4. Results**

298 The coupled model is adopted to analyze the dynamic response of the SPAR-type FOWT
 299 under combined loads from wave, wind and current. Firstly, the vortex shedding phenomenon is
 300 simulated based on 2D CFD model, and the coefficients of vortex induced loads are presented.
 301 Then, free-decay tests are conducted to show the natural characteristics of FOWT. Afterwards,
 302 four different scenarios under the rated sea state are considered, referred to as cases 1 to 4 (see,
 303 Table 3). The details of environmental parameters can be found in Table 4. In the results, the
 304 focus is on motions of the platform, including the transverse, longitudinal and vertical motions.
 305 Furthermore, some nonlinear internal resonance phenomena are observed and discussed.

306 Tab.3 Definition of load cases

Load Case	Current	Wind	Wave
1	Uniform	-	-
2	-	Steady	Irregular
3	Uniform	Steady	Irregular
4	Shear	Steady	Irregular

307
308 Tab. 4 Environmental parameters

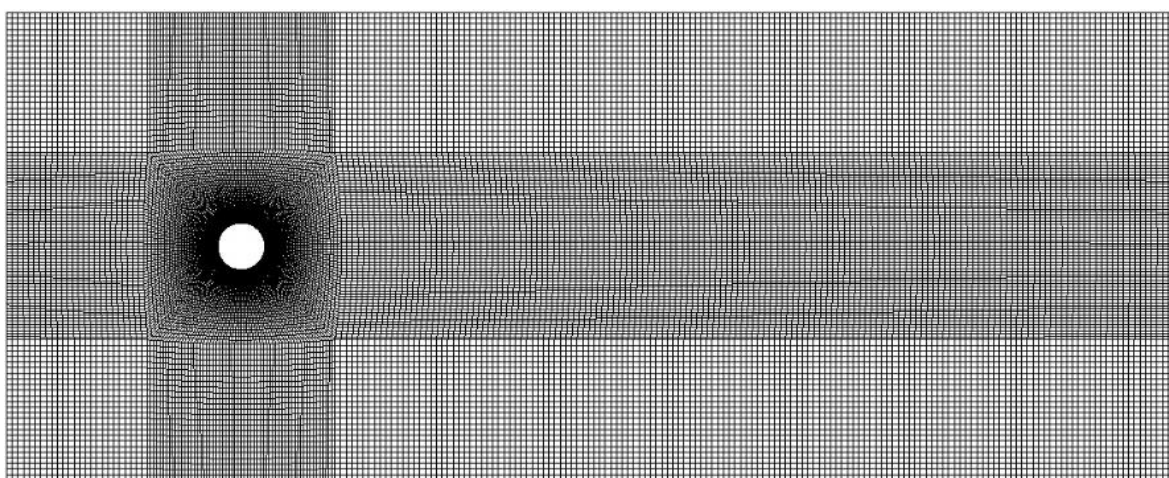
Parameter	Value
Wind speed	11.4 m/s
Surface velocity	0.6 m/s
Wave spectrum	JONSWAP
Significant wave height	6 m
Spectrum Peak Period	10s
Spectrum Peak Factor	3.3
Direction of wind, wave and current	Aligned,0°

309 Once the relative wind speed at the rotor exceeds the rated speed due to the induced velocity,
 310 the output power and aerodynamic force will increase rapidly. In order to keep the output power
 311 stable (and also for structural safety), a blade-pitch control system is necessary. In this study, a
 312 simplified quasi-static model is adopted [30]. The pitch angle is obtained *via* interpolation using
 313 the instantaneous wind speed with respect to the rotor, whose speed remains constant.

314 **4.1 Vortex shedding induced loads**

315 The configuration of our computational domain is sketched in Fig.5. According to previous
 316 research about the SPAR platform [21,38], the computational domain is a rectangular box with
 317 $-5D < x < 20D$ and $-5D < y < 5D$, where D is the diameter of the SPAR centered at (0,0). The left
 318 side of the flow area is the inflow boundary and the right side is the outflow one. Considering the

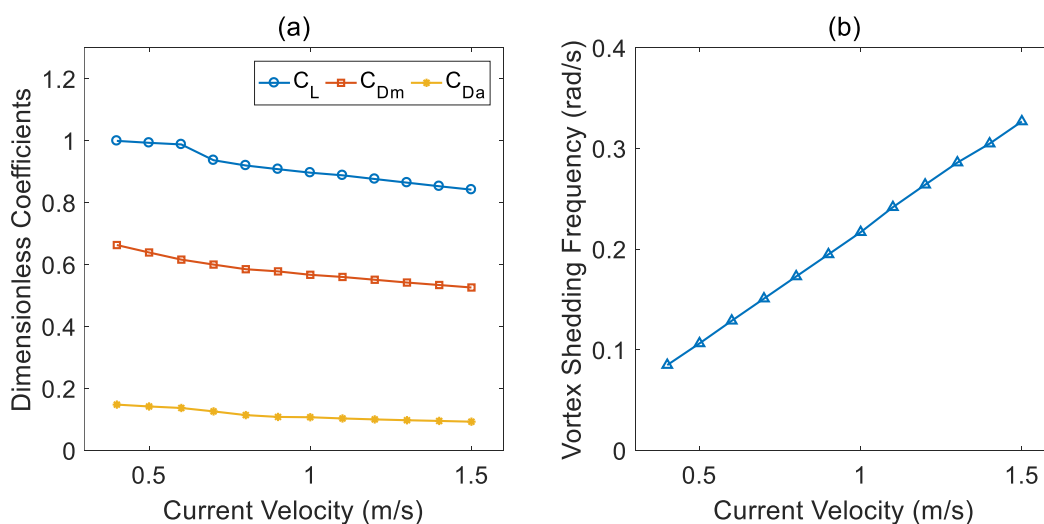
319 infinity of the flow, the upper and lower sides of the area are the symmetric boundary, and the
 320 surface of SPAR is set as non-slip boundary. The quadrilateral mesh is applied in the flow area
 321 and mesh refinements are performed in the area near the SPAR and the wake area. In the present
 322 work, we use 18250 elements in the flow field. The RNG $k-\varepsilon$ model is chosen as the turbulence
 323 model, and the standard wall functions are adopted for enhancement. Based on previous
 324 assessments on the vortex shedding effect of rigid cylinder between 2D and 3D CFD approaches,
 325 it shows that the amplitude results of 2D model are about 10% lower than the 3D model results,
 326 but the vortex shedding frequency of 2D model results shows a good quantitative agreement with
 327 the 3D one [50]. Thus, in present study, the 2D CFD approach was adopted to simulate the vortex
 328 shedding around the cylinder.



329

Fig.5 Mesh of the flow area

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Fig.6 Hydrodynamic Coefficients and vortex-shedding frequency of SPAR buoy

333 The results under different current velocities are shown in Fig.6. From the results, it can be
 334 seen that both coefficients of lift and drag loads decrease with the increasing current velocity. On
 335 the other hand, by comparing the results of C_l and C_{Da} , it is found that the amplitude of lift is
 336 almost 10 times as that of drag, but the mean values of drag load are about 60% of the amplitude
 337 of lift. That is the reason why the drag loads are often neglected or treated as steady loads in most
 338 studies on VIV or VIM.

339 In the following simulations, both the harmonic lift and drag loads will be taken into account.
 340 According to the shape of wet surface, both SPAR buoy and flow are discretized along the depth.
 341 Specifically, the profile of the current in LC4 decreases linearly with the depth, and the velocity
 342 at the seabed is assumed to be 0. At different depths, the vortex induced loads are calculated by
 343 using different velocity and vortex induced hydrodynamic coefficients based on the CFD results
 344 in Fig.6.

345 4.2 Free-decay tests

346 In order to investigate the natural frequencies of the FOWT, a series of free-decay tests are
 347 conducted. In each simulation, an initial displacement was prescribed for the corresponding DOF
 348 (1m for translational motions and 0.1rad for rotational motions) before the FOWT was released.
 349 The time histories of the motions in 6 DOFs were then recorded. According to the Fast Fourier
 350 Transformation of the time histories, the natural frequencies are shown in Table 5.

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Tab.5 Natural Frequencies of FOWT	
DOF	Natural Frequency (rad/s)
Surge	0.050
Sway	0.050
Heave	0.207
Roll	0.163
Pitch	0.163

352 Another phenomenon observed during the free-decay tests is the coupling effect between
 353 heave, roll and pitch. Fig.7 presents the results with initial heel/trim angle. The time histories of
 354 heave and roll/pitch are shown in Fig.7(a) and (b), and the response spectra of heave and
 355 roll/pitch are presented in Fig.7(c). There are 2 peaks in response of roll/pitch and 3 peaks in
 356 heave (see Table 6). According to the natural frequencies of each DOF in Table.5, we find that
 357 the frequencies of roll mode correspond to the natural frequencies of sway and roll, respectively.
 358 That is to say, roll is coupled with sway. Similarly, pitch is coupled with surge.

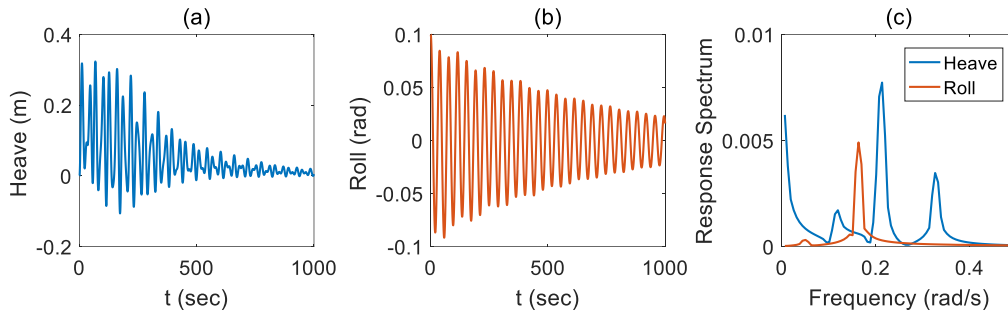


Fig.7 Coupling effects in free-decay tests of roll

Tab.6 Peak Frequencies of FOWT

DOF	Frequency (rad/s)		
	1st	2nd	3rd
Heave	0.113	0.207	0.326
Roll/Pitch	0.050	0.163	-

On the other hand, the response in heave is much more intriguing. Among the 3 peaks, the second one corresponding to the largest amplitude is the natural frequency of heave itself, and the other two correspond to super-harmonics of peak frequencies in roll. According to the nonlinear restoring force in Eq. (12), the heave mode is coupled with both roll and pitch. Based on the nonlinear dynamic theories, the second-order term of roll will induce these two-time-super-harmonic responses in heave, as the results show. The coupling effect between the DOFs will increase the response in some scenarios, and it will be further discussed later.

4.3 VIMs under the current loads (LC1)

To test the accuracy of our numerical model in simulating the VIM of SPAR-type FOWT, the responses under the current-only case (LC1) are calculated. The trajectories in the horizontal plane are shown in Fig.8 with different current velocities. It is seen that the VIMs display the figure-eight shape. These trajectories are qualitatively similar to the observed ones in model tests [15].

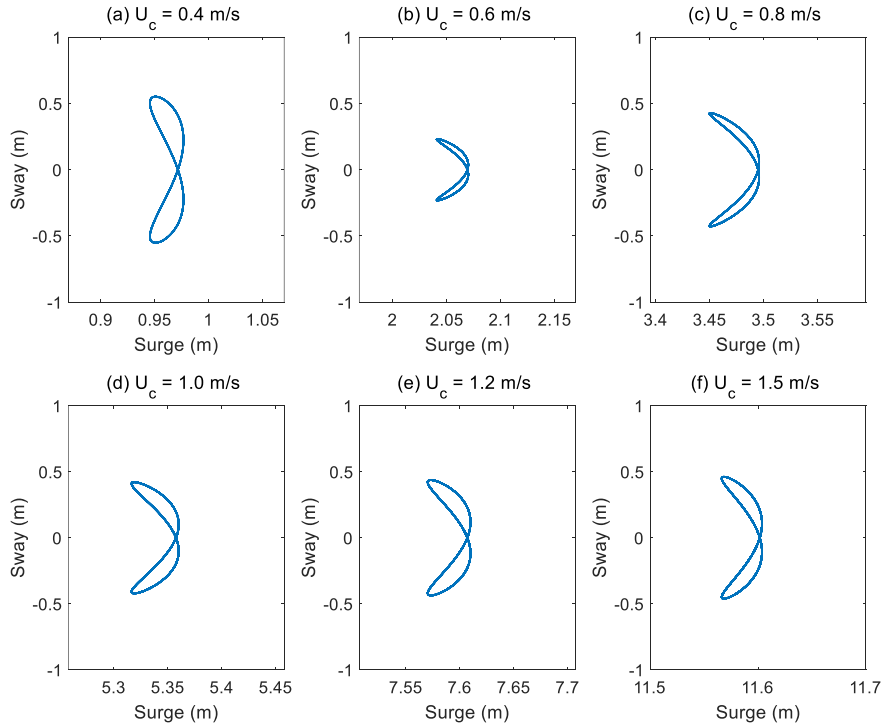


Fig.8 Trajectories of VIMs

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Fig.8 indicates that the mean displacement in surge increases with the growing current velocity, but the amplitude of oscillation remains almost unchanged. The amplitude of sway is about 0.5m, much larger than that of surge. According to the results of model tests acquired with a 1/50 model [15], the oscillation in surge is in the range from 0.015m to 0.1m, and that of sway is from 0.015m to 0.15m. The oscillations in our numerical simulations are smaller than the results in the laboratory test. In fact, the model in our simulation is based on the full-scale model. While in the model tests, both the geometric parameters and current velocity are in small-scale, in order to meet the similarity of Froude number. Thus, the Reynolds number in the test is 1/354 of that of the full-scale prototype. Moreover, the configuration of mooring system and the water depth in the model test are also different from the model in our simulation. All these differences cause quantitative difference, whereas the trajectories of VIMs in the model test are quantitatively similar to results from our numerical simulations.

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4.4 Motions under wind, wave and current

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In this section, the effect of vortex induced force on the overall responses of the system is examined. The chosen load cases are LC2(No current), LC3(Uniform flow) and LC4(Shear flow). In LC4, the current profile is discrete into several parts along the depth, and the current velocity is assumed to be uniform in each part [45]. At different depths, the vortex induced loads are calculated by using the corresponding velocity and hydrodynamic coefficients. The coefficients

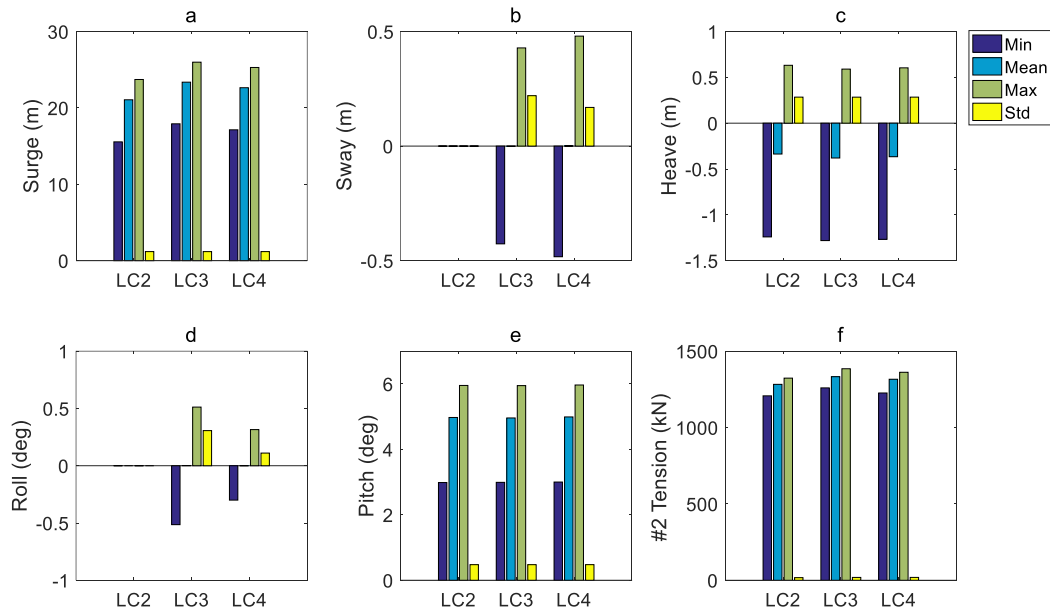
397 of vortex induced load are shown in Table 7.

398 Tab. 7 Vortex induced hydrodynamic coefficients in LC3 and LC4

Current Velocity (m/s)	Upper Bound (m)	Lower Bound (m)	Diameter (m)	C_l	C_{dm}	C_{da}	ω_s (rad/s)
LC3 Uniform Flow							
0.600	0	-14	6.5	0.988	0.616	0.137	0.182
	-14	-128	9.4	0.983	0.623	0.141	0.129
LC4 Shear Flow							
0.587	0	-14.00	6.5	0.985	0.621	0.145	0.177
0.562	-14.00	-21.33	9.4	0.981	0.623	0.141	0.122
0.550	-21.33	-40.00	9.4	0.983	0.627	0.140	0.119
0.500	-40.00	-66.67	9.4	0.993	0.639	0.142	0.106
0.450	-66.67	-93.33	9.4	0.998	0.647	0.147	0.098
0.400	-93.33	-128.00	9.4	0.999	0.663	0.148	0.085

399 The proposed 2D approach to calculated the vortex shedding in both uniform and shear is
 400 based on previous researches [46-49], and the approaches are verified with 3D simulation in
 401 those work [46,47]. Specifically, in Ref. 46, a 1/100 scale model of cell-SPAR platform with a
 402 diameter of 0.368m is chosen to perform the validation with Fluent. Both model tests and
 403 numerical simulations (including 2D and 3D) are conducted with the case where the current
 404 velocity is 0.1 m/s, including uniform and shear profile. Because of the physical limitation, the
 405 case with shear current profile is only simulated numerically. According to the results, although
 406 the 2D results are slightly larger than the 3D ones, 2D simulation meets the requirements of
 407 accuracy and efficiency. It is true that 3D-based models are physically more accurate, especially
 408 in cases with shear current profile and turbulence effects. However, these models are still too
 409 expensive. On the other hand, the results of natural period, which is one of the key mechanism to
 410 the nonlinear analysis, are more accurate in 2D simulations. Hence, this approach was adopted in
 411 our simulation.

412 In the following simulations, the overall time is 3600 and the time step is 0.1 sec. After the
 413 first 500 sec, the initial start-up transient effect has faded and the FOWT is oscillating around its
 414 dynamic equilibrium position, so the rest samples (31000) are used for statistic and FFT analysis.
 415 The results are shown in Fig.9. Among these results, three different topics are majorly discussed
 416 in the following sections, which are transvers longitudinal and vertical motions, respectively.



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Fig.9 Statistic results of motions and mooring tensions

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4.4.1 Transverse Loads and Motions

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The results of sway and roll are shown in Fig.10 and Fig.11. To distinguish the curves clearly, the logarithmic scale is adopted for the response spectra in the following subsections. Since the value of transverse loads and motions in LC2 keeps zero, they are not shown in the logarithmic spectra. Based on the time histories of lift loads in Fig. 10a and Fig.11a, multi-frequency vortex induced loads are applied on the transverse DOFs of FOWT when the current is present. Among the components in the spectra of lift loads (Fig. 10b and Fig. 11b), the responses whose frequencies correspond to the part of the buoy with a diameter of 9.4m are larger than others (Table. 7). It is attributed to the longer wet buoy of the 9.4-meter-diameter part.

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According to the statistic results of sway in Fig.9b and roll in Fig.9d, as well as the time histories in Fig.10c and 11c, it can be seen that the amplitudes in sway and roll increase significantly with the vortex induced load taken into consideration, but the mean positions keep unchanged. The response spectra of these motions are presented in Fig. 10d, 11d. It shows that the frequencies of sway and roll are in agreement with the corresponding frequencies of lift loads. These are also the vortex shedding frequencies.

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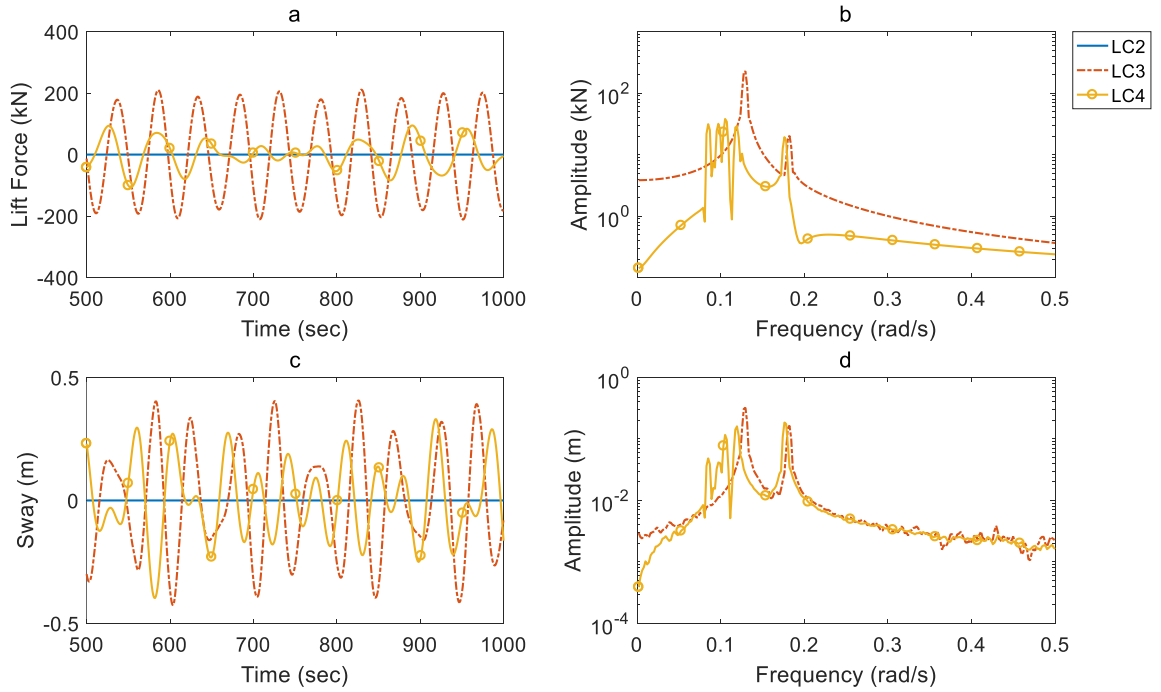
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Moreover, the effect of flow profile can be seen by comparing the results of LC3 and LC4. On one hand, there exists significant difference in the lift force between LC3 (uniform current) and LC4 (shear current). The difference in the lift moments on roll is smaller than that on sway. With these differences, both the roll and sway amplitudes in LC4 are smaller than those in LC3.

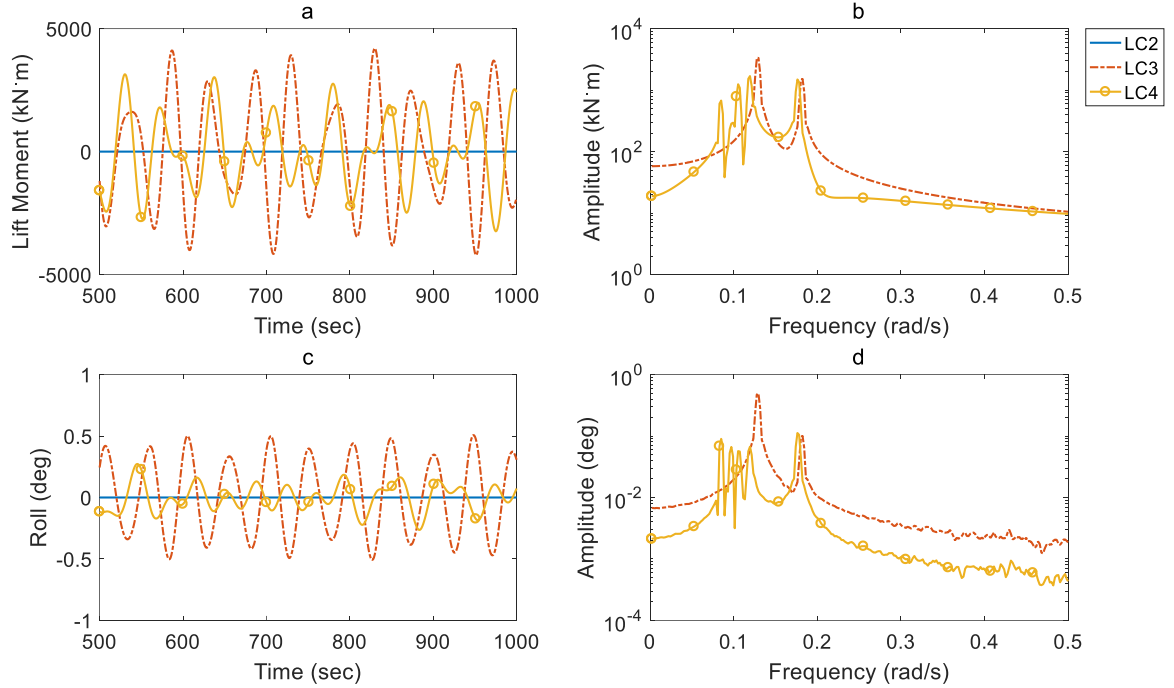
438 This is caused by the decreasing flow velocity with the increasing depth in LC4, which reduces
439 the amplitudes of lift loads. On the other hand, the frequencies of vortex shedding in LC4 are
440 more complicated because of the variation of velocity on different layers (Table. 7). This leads to
441 the occurrence of more frequencies in the transverse motion.



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Fig.10 Time histories and response spectra of sway and lift force



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Fig.11 Time histories and response spectra of roll and lift moment

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4.4.2 Longitudinal Motions

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To examine the effect of the time-varying drag loads on the longitudinal motions, in Fig.12 we show the vortex shedding loads and dynamic response in surge, and those results in pitch are presented in Fig.13. It is seen from the time histories that the mean position in surge is enlarged when the drag force is applied, while the trim angle of FOWT keeps unchanged. On the other hand, both the time histories and spectra show that the oscillations in surge and pitch are almost the same in these three cases. It leads to the graphically indistinguishable curves in the response spectra of all cases, even though the logarithmic scale is used. To summarize, unlike the transverse motions, only the mean position in surge is significantly affected by the current, other features of longitudinal motions are little changed.

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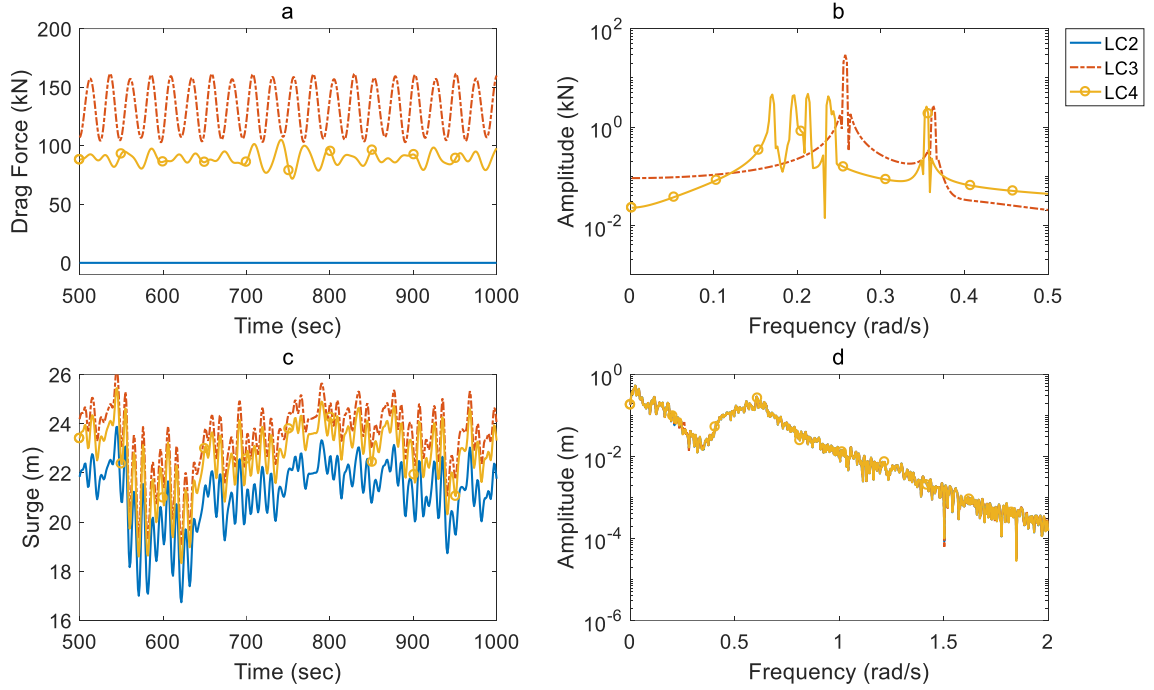
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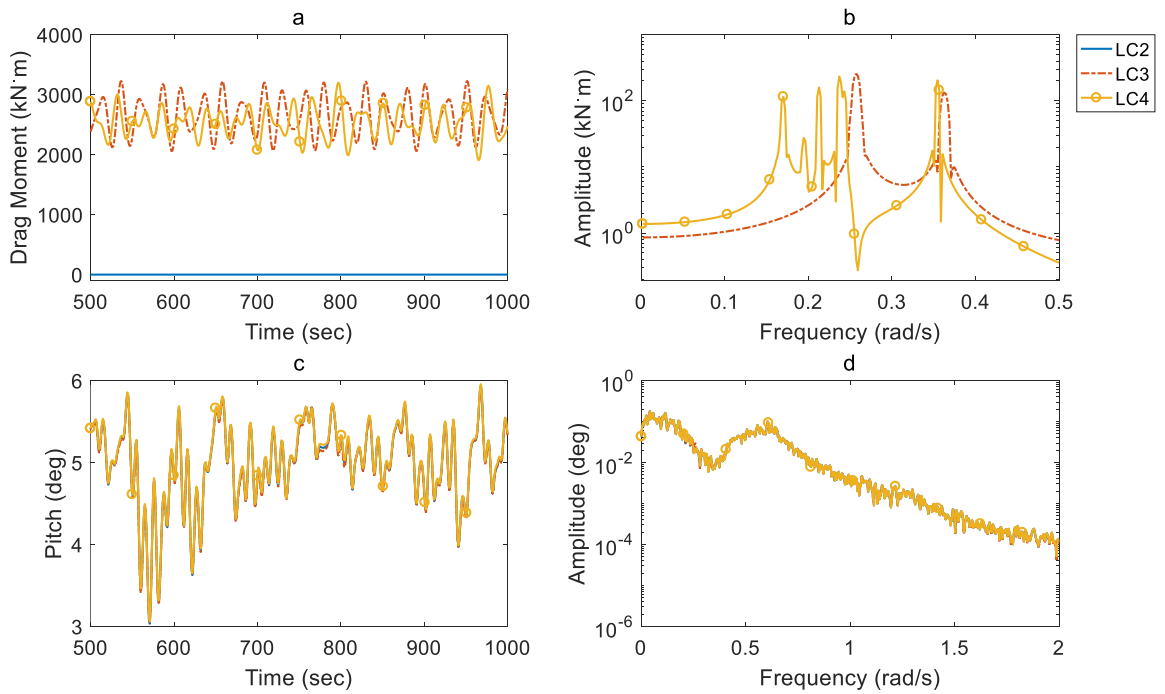
The explanation of these phenomena lies in the characteristics of the drag loads. From Fig.12a and 13a, it is seen that there exists significant difference in the mean values of drag force in different cases, and it causes the difference in the mean position under different currents. On the other hand, due to the similarity in the drag moments in pitch, the mean position in pitch is less affected. Moreover, according to the spectra, the frequencies of drag loads in LC4 is more spread than those in LC3. The oscillatory amplitude of drag is an order of magnitude less than the one of lift. As the frequencies of drag are two times those of lift, they are thus far away from the natural frequencies of surge and pitch, the oscillations in these modes are less affected.



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Fig.12 Time histories and response spectra of surge and drag force



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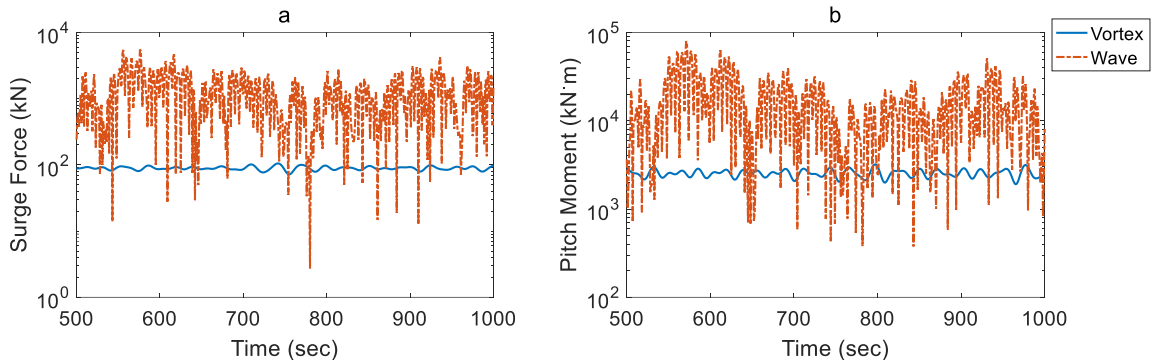
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Fig.13 Time histories and response spectra of pitch and drag moment

On the other hand, the magnitude of drag and wave loads on the platform are displayed in Fig.

469 14. According to the time histories, we found that the mean level of wave loads in surge and pitch
 470 are much larger than the vortex induced loads. In other words, the longitudinal oscillations are
 471 mostly determined by the wave forces.



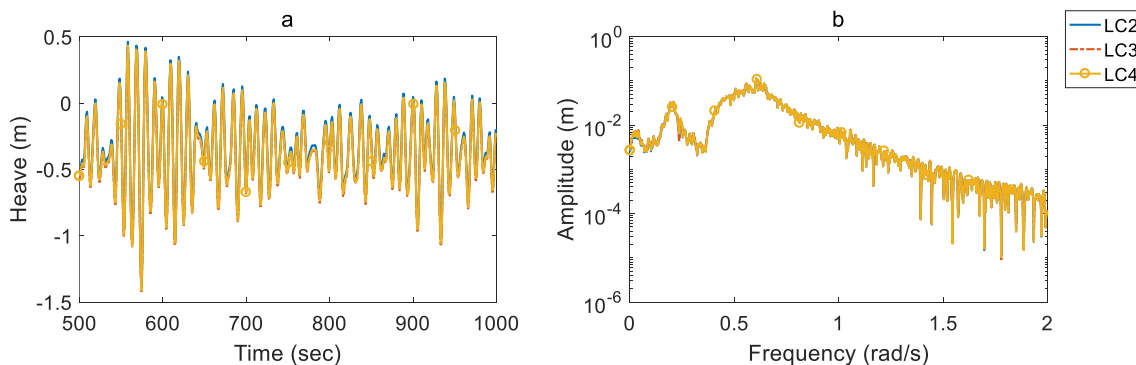
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Fig.14 Magnitude of drag and wave loads on the platform in LC3

474 **4.4.3 Vertical Motions**

475 Small difference is found in terms of the statistical results and time histories of heave in these
 476 three scenarios. Similar to the longitudinal motions, the response spectra in heave are almost
 477 identical in these cases. According to Fig. 15, although no additional load is applied on heave, the
 478 response slightly changed when the vortex excitation load is considered. This is due to the
 479 nonlinear coupling effect between heave and pitch modes. However, because the frequency of
 480 vortex shedding at the selected flow velocity is much higher than the natural frequency in heave,
 481 the current does not cause significant change in the response spectrum in heave.



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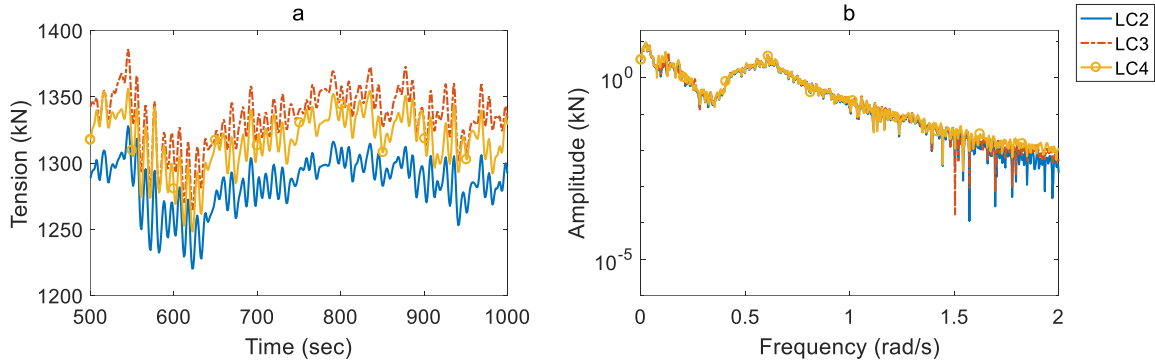
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Fig.15 Time histories and response spectrum of heave

484 **4.4.4 Tensions**

485 According to the statistic results in Fig.8 and the time histories in Fig.16a, the effect of vortex
 486 shedding on mooring tension is similar to that on the longitudinal motions. That is to say, the
 487 mean tensions are significantly increased by the current due to the increasing surge displacement.
 488 On the other hand, we found that the oscillations are similar among 3 scenarios, based on the

489 response spectra in Fig. 16b. The most significant responses are in the wave frequency, and the
490 natural frequency (in surge). Therefore, the mooring tension is determined by the surge motion
491 and incident wave, while the current has little effect on it.



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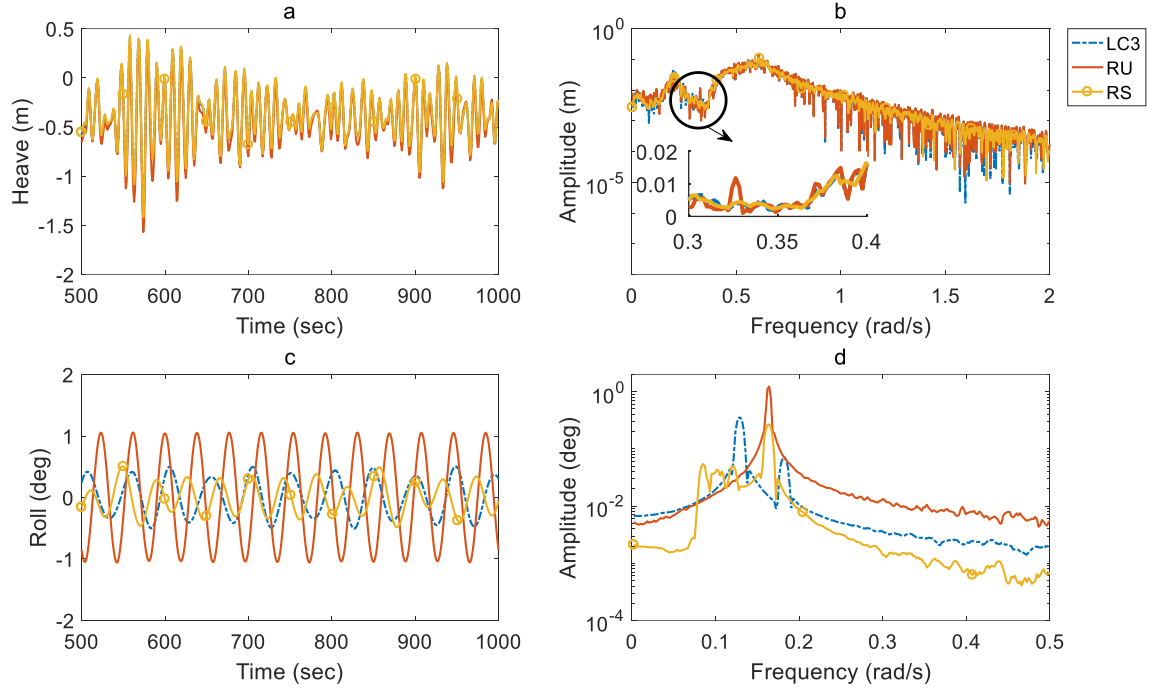
493 Fig.16 Time histories and response spectrum of tension in Line #2

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495 5. Resonance and nonlinear coupling effect

496 According to the free-decay test, there exists nonlinear coupling effect among heave, roll and
497 pitch. This effect, however, was not seen in the scenarios of rated sea states. This is attributed to
498 the fact that the wave frequency is far from the natural frequencies in these DOF so that no
499 resonance is excited. To further study the dynamic resonance and nonlinear response of the
500 FOWT, two additional load cases (hereafter referred to as cases RU and RS) in which vortex
501 shedding frequency is close to the natural frequency of roll, is examined. Based on the results of
502 free-decay tests in Table.7 and vortex shedding frequencies in Fig.6b, the current velocity is
503 chosen as 0.75m/s, corresponding to a vortex shedding frequency of 0.163 rad/s (same as the
504 natural frequency in roll). In case RU, for simplicity, the frequency is applied to both parts of the
505 buoy, and the flow is assumed to be uniform. On the other hand, a shear profile is used in case
506 RS, which means that a portion of the profile has the current velocity corresponding to the
507 shedding frequency close to the natural frequency in roll. Case 3 is chosen for comparison.

508 The motions in heave and roll are presented in Fig.17. Large amplitude of roll occurs in case
509 R at the vortex shedding frequency (same as natural frequency in roll). Besides, a new peak at
510 0.3243 rad/s appears in the response spectrum of heave (Fig.17b). It is twice the vortex shedding
511 frequency. This super-harmonic response is caused by the nonlinearly coupled restoring forces in
512 heave and roll. Compared with the uniform flow case (Case RU), the resonant shear flow (Case
513 RS) induces multi-frequency responses in roll, but the amplitude is much less than that of Case
514 RU. On the other hand, the super-harmonic effect is not pronounced in the heave response of
515 Case RS. This is due to the small amplitude of resonant response in roll.



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Fig.17 Resonance in roll and nonlinear coupling effect in heave

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Therefore, when the wave, wind and current are all considered, the motion of SPAR-type FOWT may be affected significantly by vortex induced loads, especially in cases with resonance. In these cases, the amplified motion becomes a source of parametric excitation to affect responses in other DOF. For example, as illustrated above, due to the nonlinear coupling between heave and roll, resonance in roll induces super-harmonic (twice the vortex shedding frequency) response in heave. If this super-harmonic frequency happens to be close to the natural frequency in heave, it will in turn generate large response in heave. This is the internal resonance scenario found in SPAR-type offshore platforms [33].

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6. Conclusions

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A coupled dynamic model has been developed in time domain to investigate the dynamic responses of SPAR-type FOWT under the combined sea state of wind, wave and current. It includes a blade-element-momentum model of rotating blades, a nonlinear coupling hydrostatics model of floating structure, a three-dimensional nonlinear model of the free-surface effects on the SPAR buoy, a mooring model based on catenary theory, and a vortex-shedding model using computational fluid dynamics.

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Based on this model, the motion of FOWT under the combined effects of vortex shedding, wave excitation, aerodynamic load and mooring load is calculated in time domain. The influence of both uniform current and shear current is analyzed. After the current is included, the vortex

536 shedding will induce both lift loads and drag loads. The lift force and moment act directly on the
537 sway and roll, causing transverse motions with the vortex shedding frequency. The drag force
538 and moment are applied on the motions in the longitudinal direction. The mean position of surge
539 is increased by the drag force. The oscillation in surge and the pitch motion are not significantly
540 affected.

541 Compared with the uniform current, depth-dependent velocities of the shear current lead to
542 variations of vortex shedding frequencies along the platform, which makes the frequencies of
543 transverse motions more diversified. With the water depth increasing, the flow velocity decreases
544 gradually, so do lift and drag. Therefore, the transverse motion caused by the shear current is
545 smaller than that caused by the uniform flow with the same surface velocity. However, the flow
546 profiles have no significant effect on longitudinal motion.

547 Due to the nonlinear coupling effect, the resonant lift load may affect not only roll but also
548 other DOFs. For example, when the vortex shedding frequency is close to the natural frequency
549 in roll, in addition to large resonance response in roll, large motion may also appear in heave at a
550 super-harmonic frequency.

551 Although our simulations are conducted by using a specific design as an example, the results
552 may have much broader implications. Firstly, these numerical studies suggest that the effects of
553 current could be important for the response of FOWT. Therefore, such environmental condition
554 should be taken into account during the design process. Moreover, both the lift and drag loads
555 may cause responses in all DOFs. Neither of them should be neglected in the simulation. Finally,
556 the coupling effect may create super-harmonic responses in heave mode, even leading to internal
557 resonance at certain ratios of frequencies. These issues may cause potential dangers or damages.

558 It is necessary to point out that in the present work the interactions between wind and blades
559 are modeled *via* the BEM theory. Though this classical algorithm has been widely used, the
560 dynamic and unsteady effects are not included. To accurately simulate these effects, a dynamic
561 model (e.g. three-dimensional potential theory [34,35]) may be adopted in the following study.

562 Besides, the vortex shedding is calculated *via* a 2D CFD model without considering the 3D
563 effect. This may lead to inaccuracies in the prediction [36]. Future investigations about the 3D
564 effect are required. Moreover, in this work we focus on resonance in roll, whereas the potential
565 resonance in sway is not considered (as illustrated in our CFD simulation, in our particular case
566 the vortex shedding frequency is much larger than the natural frequency in sway). However, in
567 certain scenarios resonance in sway is observed [15]. Further study on this issue is also needed.

568 In reality, structural vibrations may affect the dynamic response as well as the environmental
569 loads on the structure, especially for slender bodies such as mooring cables, blades and the tower.
570 In the present study, the flexibility of these structures is not included. To analyze these issues
571 more thoroughly, further simulations about the aero-elastic or hydro-elastic effects are needed.

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579

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