

KU-KI-885		No. TCD-4640-97
<u>Date August 21, 1997</u>		
<p>Japan Civil Aviation Bureau</p> <p>TAIKUSEI-KAIZEN-TSUHO</p> <p><u>Airworthiness Directive</u></p> <p>The undermentioned examinations or modifications are mandatory.</p>		
1. Applies to : Kawasaki Hughes Model 369HS and 369D, and Hughes Model 369HS and 369D helicopters.		
2. Compliance required as indicated, unless already accomplished. To prevent the fatigue crack(s) of main rotor blade attaching pin, accomplish the following. 2.1 Within 10 flight hours after the effective date of this AD, inspect main rotor blade attaching pin and damper attaching pin, in accordance with Kawasaki Service Bulletin (KSB) No. KSB-369-334 dated July 14, 1997, or KSB No. KSB-369D-213 dated July 14, 1997, as applicable.		

Japan Civil Aviation Bureau	No. TCD-4640-97	sheet 2/2
<p>2.2 If any discrepancies are found by the inspection required by paragraph 2.1, replace the pin and, if necessary, apply a corrosion-prevention treatment, in accordance with KSB No. KSB-369-334 or KSB No. KSB-369D-213, as applicable.</p> <p>2.3 Alternate means of compliance with this AD may be used if approved by Director-General of Civil Aviation Bureau.</p>		
<p>3. Remarks</p> <p>3.1 This AD becomes effective on August 28, 1997.</p> <p>3.2 Kawasaki Service Bulletin No. KSB-369-334 dated July 14, 1997, No. KSB-369D-213 dated July 14, 1997, or JCAB-approved later revisions pertain to this subject.</p>		